

KRILA

1. Avion ima eliptična krila površine 20 m^2 i raspona 10 m izrađena od ravne ploče. Promatra se let na visini 2000 m brzinom 400 km/h pod geometrijskim napadnim kutom od 4° . Odrediti:

- kako se mijenja kut inducirane brzine za eliptično krilo,
- silu uzgona,
- silu otpora,
- snagu potrebnu za savladavanje induciranog otpora.

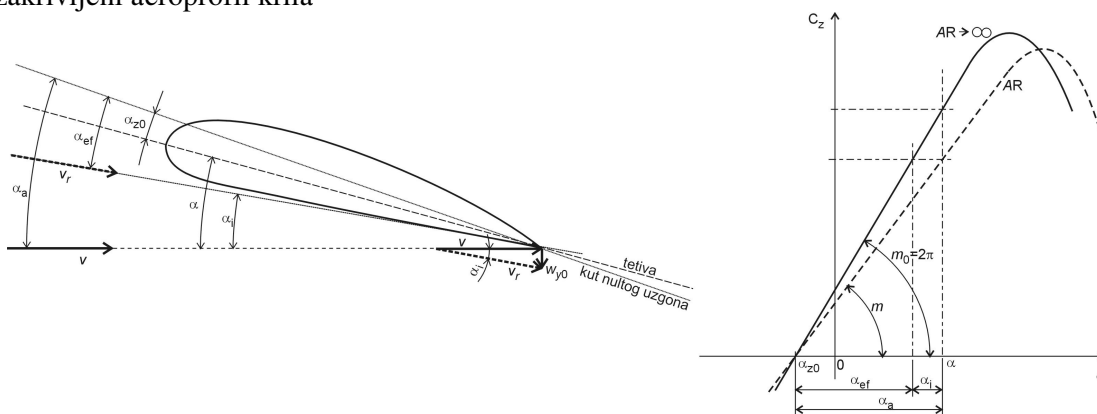
Napomena: smatrati da su lokalne vrijednosti koeficijenta uzgona za aeroprofil u obliku ravne ploče jednake idealnoj vrijednosti: $c_z = 2 \cdot \pi \cdot \alpha$.

Rješenje: $H = 2000 \text{ m} \rightarrow \rho = 1,00646 \text{ kg/m}^3$

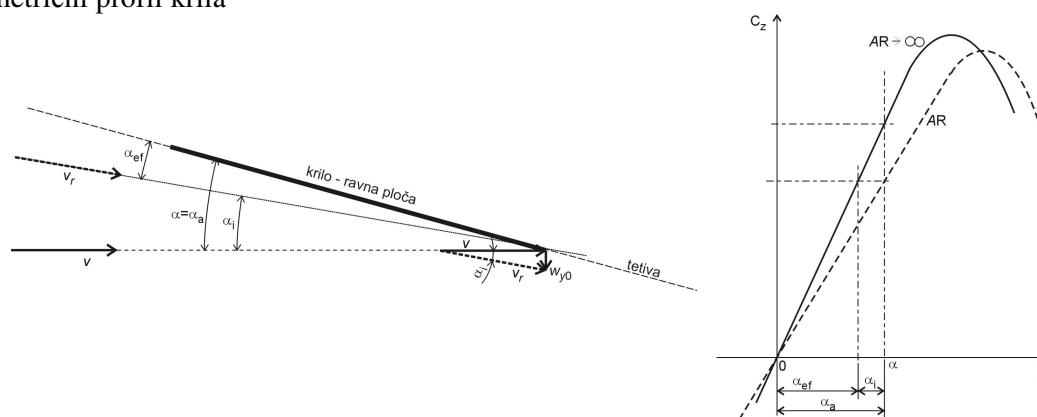
$$v = 400 \text{ km/h} = 111,1 \text{ m/s}$$

$$S = 20 \text{ m}^2; \quad b = 10 \text{ m} \rightarrow AR = \frac{b^2}{S} = \frac{10^2}{20} = 5$$

– zakrivljeni aeroprofil krila



– simetrični profil krila



$$a) \quad AR \rightarrow \infty \rightarrow c_z = m_0 \cdot \alpha_a = 2 \cdot \pi \cdot \alpha_a \quad (m_0 = 2 \cdot \pi)$$

$$AR \rightarrow m = \frac{m_0}{1 + \frac{m_0}{\pi \cdot AR}} = \frac{2 \cdot \pi}{1 + \frac{2 \cdot \pi}{\pi \cdot AR}} = \frac{2 \cdot \pi \cdot AR}{2 + AR} = \frac{2 \cdot \pi \cdot 5}{2 + 5} = 4,488$$

$$C_z = m \cdot \alpha_a = 4,488 \cdot \alpha_a$$

$$\alpha_i = \frac{C_z}{\pi \cdot AR} = \frac{4,488 \cdot \alpha_a}{\pi \cdot 5} = 0,2857 \cdot \alpha_a$$

$$\text{b) } \alpha_a = \alpha = 4^\circ = 4 \cdot \frac{\pi}{180} \text{ rad}$$

$$C_z = 4,488 \cdot \alpha_a = 4,488 \cdot 4 \cdot \frac{\pi}{180} = 0,3133$$

$$Z = C_z \cdot \frac{1}{2} \cdot \rho \cdot v^2 \cdot S = 0,3133 \cdot \frac{1}{2} \cdot 1,00646 \cdot 111,1^2 \cdot 20 = 38929 \text{ N}$$

$$\text{c) } C_{xi} = \frac{C_z^2}{\pi \cdot AR} \cdot (1 + \delta) = \frac{C_z^2}{\pi \cdot e \cdot AR}$$

$$e = \frac{1}{1 + \delta} \text{ - Oswaldov koeficijent (za eliptično krilo: } e = 1; \delta = 0)$$

$$C_{xi} = \frac{C_z^2}{\pi \cdot AR} = \frac{0,3133^2}{\pi \cdot 5} = 0,00625$$

$$X_i = C_{xi} \cdot \frac{1}{2} \cdot \rho \cdot v^2 \cdot S = 0,00625 \cdot \frac{1}{2} \cdot 1,00646 \cdot 111,1^2 \cdot 20 = 776,4 \text{ N}$$

$$\text{d) } P_i = X_i \cdot v = 776,4 \cdot 111,1 = 86272 \text{ W} \approx 86,3 \text{ kW}$$

$$\underline{\text{II naćin:}} \quad \alpha_i = 0,2857 \cdot \alpha_a = 0,2857 \cdot 4 = 1,1428^\circ$$

$$\alpha_{ef} = \alpha_a - \alpha_i = 4 - 1,1428 = 2,8572^\circ$$

$$C_z = m_0 \cdot \alpha_{ef} = 2 \cdot \pi \cdot 2,8572 \cdot \frac{\pi}{180} = 0,3133$$